1	Aircraft Measurements of Gravity Waves in the Upper Troposphere and Lower
2	Stratosphere during the START08 Field Experiment
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#### Abstract

24 This study analyzes *in-situ* airborne measurements from the 2008 Stratosphere-Troposphere 25 Analyses of Regional Transport (START08) experiment to characterize gravity waves in the 26 extratropical upper troposphere and lower stratosphere (ExUTLS) region. The focus is on the 27 second research flight (RF02), which took place on 21-22 April 2008. This was the first airborne 28 mission dedicated to probing gravity waves associated with strong upper-tropospheric jet-front 29 systems. Based on spectral and wavelet analyses of the *in-situ* observations, along with a 30 diagnosis of the polarization relationships, clear signals of mesoscale variations with 31 wavelengths ~50-500 km are found in almost every segment of the 8-hr flight, which took place 32 mostly in the lower stratosphere. The aircraft sampled a wide range of background conditions 33 including the region near the jet core, the jet exit and over the Rocky Mountains. In contrast to 34 the long wavelength mesoscale variations, smaller-scale wavelike oscillations below 50 km are 35 found to be quite transient. In particular, aircraft measurements of several flight segments are 36 dominated by signals with sampled periods of ~20-~60 seconds and wavelengths of ~5-~15 km 37 (assuming that the typical flight speed is approximately 250 m/s). We speculate that at least part 38 of these nearly-periodic high-frequency signals are a result of intrinsic observational errors in the 39 aircraft measurements or small-scale flight-altitude fluctuations that are difficult to fully 40 characterize. Despite the presence of possibly spurious wave oscillations in several flight 41 segments, the power spectra of horizontal winds and temperature averaged over the analyzed 42 START08 flight segments generally follow the -5/3 power law.

44 **1. Introduction** 

45 One of the challenges to understanding the extratropical upper troposphere and lower 46 stratosphere (ExUTLS) is that dynamical processes with a wide range of scales occur in the 47 region. Gravity waves, in particular, are known to play a significant role in determining the 48 structure and composition of the ExUTLS. Tropopause jets and fronts are significant sources of 49 gravity waves (O'Sullivan and Dunkerton 1995; Reeder and Griffins 1996; Zhang 2004; Wang 50 and Zhang 2007; Mirzaei et al. 2014; Wei and Zhang 2014, 2015), along with surface 51 topography (Smith 1980) and moist convection (Lane et al. 2001). Gravity waves above the jet 52 may be responsible for double or multiple tropopauses (Yamanaka et al. 1996; Pavelin et al. 53 2001) and may contribute to layered ozone or PV structures (Bertin et al. 2001). Also, strong 54 horizontal and vertical shear in the layer and the discontinuity in static stability at the tropopause 55 provide a favorable environment to reflect, capture, break and dissipate gravity waves generated 56 in the lower troposphere, such as those produced by surface fronts (Plougonven and Snyder 57 2007). Gravity wave breaking and wave-induced turbulence (e.g., Koch et al. 2005) can 58 contribute significantly to mixing of trace gases in the ExUTLS, thereby affecting chemical 59 composition (Vaughan and Worthington, 2000). Also, convectively-generated gravity waves 60 may extend the impact of moist convection far above cloud tops through wave-induced mixing 61 and transport (Lane et al. 2004).

In particular, mesoscale gravity waves with horizontal wavelength of ~50-~500 km are known to occur in the vicinity of unbalanced upper-tropospheric jet streaks and on the cold-air side of surface frontal boundaries (Uccellini and Koch 1987; Plougonven and Zhang 2014). This phenomenon has been identified repeatedly in both observational studies (Uccellini and Koch 1987; Schneider 1990; Fritts and Nastrom 1992; Ramamurthy et al. 1993; Bosart et al. 1998;

67 Koppel et al. 2000; Rauber et al. 2001; Plougonven et al. 2003) and numerical investigations of 68 the observed cases (Powers and Reed 1993; Pokrandt et al. 1996; Kaplan et al. 1997; Zhang and 69 Koch 2000; Zhang et al. 2001, 2003; Koch et al. 2001, 2005; Lane et al. 2004). In addition, 70 idealized simulations of dry baroclinic jet-front systems in a high-resolution mesoscale model 71 have been performed to investigate the generation of mesoscale gravity waves (Zhang 2004), the 72 sensitivity of mesoscale gravity waves to the baroclinicity of jet-front systems (Wang and Zhang 73 2007), and the source of gravity waves with multiple horizontal scales (Lin and Zhang 2008). 74 Most recently, Wei and Zhang (2014, 2015) studied the characteristics and potential source 75 mechanisms of mesoscale gravity waves in moist baroclinic jet-front systems with varying 76 degree of convective instability.

77 Advances in space technology provide the means to observe gravity waves in detail. 78 Recent studies have demonstrated that satellites such as Microwave Limb Sounder (MLS) and 79 Advanced Microwave Sounding Unit-A (AMSU-A) offer quantitative information of gravity 80 waves in the middle atmosphere (Alexander and Rosenlof 2003; Wu and Zhang 2004; Zhang et 81 al. 2013). In addition to satellite measurements, gravity waves are also observed by surface 82 observations (Einaudi et al. 1989; Grivet-Talocia et al. 1999; Koppel et al. 2000), high-resolution 83 radionsonde networks (Vincent and Alexander 2000; Wang and Geller 2003; Zhang and Yi 84 2007; Gong and Geller 2010), radars (Vaughan and Worthington 2000, 2007), and super-85 pressure balloons (Hertzog and Vial 2001).

Among the abovementioned observational tools, aircraft have also been widely used as *in-situ* measurements of gravity waves. Probably since Radok (1954), which was one of the first observations of mountain waves with aircraft, past aircraft field campaigns have mainly focused on terrain-induced gravity waves (Radok 1954; Vergeiner and Lilly 1970; Lilly and Kennedy

90 1973; Smith 1976; Karacostas and Marwitz 1980; Brown 1983; Moustaoui et al. 1999; 91 Leutbecher and Volkert 2000; Poulos et al. 2002; Dornbrack et al. 2002; Doyle et al. 2002; 92 Smith et al. 2008). The recent Terrain-Induced Rotor Experiment (T-REX) in March-April 2006 93 (Grubišić et al. 2008) was the first full research project to use the National Science Foundation 94 (NSF) – National Center for Atmospheric Research (NCAR) Gulfstream V (GV) (Laursen et al. 95 2006), which has better Global Positioning System (GPS) accuracy than the previous versions. 96 The National Aeronautics and Space Administration (NASA) high-altitude ER-2 research 97 aircraft was also employed during the recent Cirrus Regional Study of Tropical Anvils and 98 Cirrus Layers Florida Area Cirrus Experiment (CRYSTAL-FACE) (Jensen et al. 2004), which 99 conducted research flights in the vicinity of sub-tropical and tropical deep convection to study 100 the effects of convectively generated gravity waves (Wang et al. 2006). However, systematic in-101 situ measurements of mesoscale gravity waves, especially those associated with upper-102 tropospheric jet-front systems in the ExUTLS are very scarce. Relevant work includes Nastrom 103 and Fritts (1992) and Fritts and Nastrom (1992), who used commercial aircraft measurements to 104 infer the different sources of gravity waves (convections, front, topography, and jet streaks). 105 They found that mesoscale variances of horizontal wind and temperature were large at the jet-106 front vicinity regions. However, little is known quantitatively about the generation mechanisms, 107 propagation and characteristics of gravity waves associated with the tropospheric jet streaks. 108 This is due in part to the fact that gravity waves are transient in nature and hard to resolve with 109 regular observing networks (Zhang et al. 2004).

The recent Stratosphere-Troposphere Analyses of Regional Transport 2008 (START08)
experiment was conducted to examine the chemical structure of the ExUTLS in relation to
dynamical processes spanning a range of scales (Pan et al. 2010). In particular, one specific goal

113 of START08 was to observe the properties of gravity waves generated by multiple sources, 114 including jets, fronts, and topography. During the START08 field campaign, a total of 18 115 research flight (RF) missions were carried out during April-June 2008 from the NCAR aviation 116 facility in Broomfield, Colorado (also see the online field catalog of the 18 RFs at 117 http://catalog.eol.ucar.edu/start 08/missions/missions.html). The second flight (RF02), which 118 occurred on 21-22 April 2008, was dedicated, to our knowledge for the first time, to probing 119 mesoscale gravity waves associated with a strong upper-tropospheric jet-front system, even 120 though some previous studies may have recognized the presence of these waves (e.g., Shapiro 121 and Kennedy 1975; Koch et al. 2005). Although only one flight specifically targeted gravity 122 waves, many of the other flights during START08 obtained high-quality observations of gravity 123 waves in the ExUTLS under a wide range of meteorological conditions. This study is an analysis 124 of the gravity wave observations from the START08 mission.

A brief description of the experimental design for RF02 and its corresponding meososcale simulation are presented in section 2, followed in section 3 by a review of the flightlevel measurements. Section 4 investigates the localized wave variance with wavelet analysis and examines the polarization relationship based on cospectrum/quadraspectrum analysis. Several examples of wave-like variances are shown and discussed in section 5. Section 6 contains a summary.

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# 132 **2.** Experimental design

The GV research aircraft is ideally suited for investigating gravity waves in the ExUTLS region. The flight ceiling of the aircraft is about 14 km with the START08 payload, which enables sampling the vertical structure of the ExUTLS. With a typical flight speed of ~250 m/s at

136 cruise altitude, the flight duration of  $\sim 8$  hours for a single flight enables the GV to sample a large 137 geographic area with high-resolution (1-Hz) in-situ observations. A total of 68 flight segments 138 (color lines in Fig. 1) during the START08 are selected for analysis (also see Fig. 2 in Pan et al. 139 2010 for GV ground tracks of the 18 RFs). Each of these flight segments is longer than 200 km 140 and has near-constant flight-level static pressure and a relatively straight path. This will largely 141 eliminate spurious wave variance due to rapid changes in direction or altitude. In particular, the 142 RF02 mission was conducted over the central United States (38.87-51.10°N, 94.00-109.95°W) to 143 study the gravity wave excitation from a jet-front system and topography in the ExUTLS (Fig. 2 144 and Table 1). It started at 17:53 UTC on 21 April 2008 and finished at 02:54 UTC on 22 April 145 2008. This ~8-hour flight covered a total horizontal distance of ~6700 km, mostly in the lower 146 stratosphere. Five flight segments (thick blue lines in Fig. 1; thick blue lines in Fig. 2b-Fig. 2f; 147 details in section 3) in RF02 are used here. For most of the 5 flight segments, the aircraft flew at an altitude of ~12.5 km (red lines in Fig. 3d; Table 1) and at a speed of ~250 ms<sup>-1</sup> (Table 1). 148

149 The Weather Research and Forecast (WRF) model (Skamarock et al. 2005) was used for 150 flight-planning forecasts. Real-time forecasts used WRF version 2.2.1 and were run with 45-km 151 and 15-km grid spacing for single deterministic forecasts (D1 and D2 in Fig. 1) and 45-km grid 152 spacing for ensemble prediction (D1 only). The model was initialized with a 30-member 153 mesoscale ensemble-based multi-physics data assimilation system (Zhang et al. 2006; Meng and 154 Zhang 2008a,b) and assimilated standard radiosonde observations. The real-time WRF forecasts 155 were archived at the START08 field catalog (http://catalog.eol.ucar.edu/cgi-156 bin/start08/model/index). The flight track of RF02 was assigned to fly across the jet exit region 157 and gravity wave active area predicted by the real-time forecasts (also see Fig. 11 in Pan et al. 158 2010 for the real-time mesoscale forecast of gravity waves). Higher-resolution post-mission

159 WRF simulations with 5-km and 1.67-km grid spacing (D3 and D4 in Fig. 1) were also 160 conducted to examine the role of small-scale dynamical processes (e.g., convection and gravity 161 waves), which will be briefly reported in section 3. Nevertheless, an in-depth investigation of the 162 gravity wave dynamics based on the high-resolution post-mission WRF simulations is beyond 163 the scope of the current study, and will be reported elsewhere.

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### 3. Overview of the flight-level measurements

166 Figure 2 depicts the track design of the entire flight and five flight segments during RF02, 167 along with the horizontal wind speed and the smoothed horizontal divergence near the flight 168 level simulated by the high-resolution post-mission WRF simulations valid at different 169 representative times of each five segments. Three flight segments pass mainly along an upper-170 tropospheric jet streak. These are labeled J1, J2, and J3 and are displayed in Fig. 2b, 2c, and 2d, 171 respectively. Two other flight segments cross the mountains and high plains of Colorado and 172 Kansas. These are labeled M1 and M2 and are displayed in Fig. 2e and 2f, respectively. Flight 173 segment J3 is the longest during RF02. That segment includes flight through or above: the jet 174 core (gray shading in Fig. 2), a jet over high mountains (see the terrain map in Fig. 1), the exit 175 region of the jet, and a surface cold front (not shown). The other two segments, J1 and J2, were 176 intended to be a single segment, but an altitude change was necessary due to air traffic control.

Guided by the WRF model forecasts (e.g., Fig. 11 in Pan et al. 2010), this GV flight 177 178 mission sampled WRF-predicted gravity waves with different potential sources including 179 imbalance of jet streak and orographic forcing. Figure 3 shows the along-track horizontal 180 velocity component (u), across-track horizontal velocity component (v), horizontal wind speed  $(V; V = \sqrt{u^2 + v^2})$ , vertical velocity component (w), potential temperature ( $\theta$ ), corrected static 181

182 pressure  $(p_c)$ , static pressure  $(p_s)$ , hydrostatic pressure correction  $(p_h)$  derived from the airborne 183 *in-situ* measurements as well as flight height, and terrain along each of the five flight segments. 184 To facilitate spectral and wavelet analyses of these measurements, each variable from the 1-Hz 185 aircraft measurement along the flight segment is linearly interpolated into 250-m spatial series 186 with fixed resolution in distance. The right-hand rule is used to determine the relationships 187 among the positive along-track directions, the positive across-track directions, and the positive 188 vertical directions. For segments J1, J2, and J3, the positive along-track (across-track) directions 189 are all approximately toward the northeast (northwest). For segments M1 and M2, the positive 190 along-track (across-track) directions are both approximately toward the east (north). The corrected static pressure  $p_c$  is calculated using the formula of Smith et al. (2008, their equation 191 192 12):

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$$p_c = p_s + p_h = p_s + \bar{\rho}g(z - z_{ref}) \quad (1)$$

where z is the GPS altitude,  $z_{ref}$  is the average altitude of flight segment and  $\bar{p}$  is the average density of flight segment. Corrected static pressure  $p_c$  from equation 1 is to correct the measured static pressure  $p_s$  to a common height level (i.e.,  $z_{ref}$ ) based on the assumption of local hydrostatic balance. Smith et al. (2008) suggests that the contribution of  $p_s$  to  $p_c$  is much smaller than  $p_h$ , because it is assumed that the aircraft almost flies on an isobaric surface.

Consistent with what was predicted by the real time WRF forecast guidance (as shown in Fig.11 of Pan et al. 2010) as well as simulated by the high-resolution post-mission WRF simulations (in particular the horizontal divergence as potential signals of gravity waves as shown in Fig. 2), the GV *in-situ* measurements of different atmospheric variables suggest there are prevalent gravity wave activities along almost every leg of the 8-hr flight, most notably in the vertical motion field. The largest amplitude of *w* (over 2 m/s) is during the middle portion of 205 segment J3 (location 680-780 km) on the lee slopes of the Rocky Mountains (also see the 206 discussion in section 5.2). The high terrain and the lee slopes also have the enhanced vertical 207 motions for both segment M1 and segment M2. Though not as large in amplitude, enhanced 208 fluctuations of vertical motions are also observed in the northern end of segment J3, which is in 209 the exit region of the upper-level jet streak and above the surface front. The enhanced variances 210 of vertical motion, accompanied by the changes in horizontal wind and potential temperature, 211 may be associated with topography for both M1 and M2 segments, even though the role of jet 212 cannot be isolated.

213 Power spectra of five selected aircraft measurement variables are given in Fig. 4 for each 214 of the five flight segments during RF02. The calculations of the spectra are performed with the 215 "specx anal" function in the NCAR Command Language (NCL). Several steps are done before 216 the calculations. Firstly, the mean and least squares linear trend in each of the series are 217 removed. Secondly, smoothing by averaging 7 periodogram estimates is performed. Thirdly, 10% of the series are tapered. For segment J1, u, v,  $\theta$  and  $p_c$  have several significant spectral 218 219 peaks for wavelengths ranging from 16-128 km (mesoscales). The statistically significant 220 spectral peaks in w are more for smaller scales, one at 2-4 km, and the other at 8-32 km. The 221 spectral characteristics for segment J2 are mostly the same as J1 except for much less power at 222 longer wavelengths (16-128 km) and only one peak at smaller scales (2-8 km). For segment J3, 223 both u and  $\theta$  have statistically significant spectral peaks at mesoscales (~50 and 128 km) and at 224 smaller scales (8-16 km), the later (not the former) of which is also very pronounced for the w spectrum. No significant spectral peak is found for the corrected static pressure  $p_c$  for segment 225 226 J3, except at 512 km, which is likely a reflection of the sub-synoptic scale pressure patterns at 227 the flight level (Fig. 2d). For segment M1, there is a significant mesoscale spectral peak at around 32-64 km for u,  $\theta$  and  $p_c$ , while smaller-scale variations from 4-16 km are also significant for nearly all variables except for  $p_c$ . There are almost no significant spectral peaks for all 5 variables for segment M2 except for around 2 km for w.

231 Past studies from both aircraft observations (e.g., Nastrom and Gage 1985; Bacmeister et 232 al. 1996; Lindborg 1999) and numerical simulations (e.g., Skamorcok 2004; Waite and Snyder 233 2013) have revealed/verified the existence of an approximate -5/3 power law that is expected for 234 the direct energy cascade in isotropic three-dimensional turbulence (e.g., Kolmogorov 1941) and the inverse cascade in two dimensions (e.g., Kraichnan 1967), as well as an approximate -3 235 236 power law that is expected for quasigeostrophic turbulence theory (e.g., Charney 1971). The 237 spectral slopes of different variables derived from the flight-level measurements from START08 238 are thus examined here in detail. Overall in segment J3, the spectrum slope for  $\theta$  (the third 239 column in Fig. 4d) is remarkably similar to those for u (the third column in Fig. 4a) and v (the 240 third column in Fig. 4b), except that there appears to be a deviation from both -3 and -5/3 power 241 laws for scales of  $\sim 8-\sim 16$  km. The spectral slope of w (the third column in Fig. 4c) is also similar 242 to that of  $\theta$  (the third column in Fig. 4d) for all scales below 32 km, including the above-243 mentioned deviation. However, for scale larger than  $\sim$ 32 km, the slope of w (the third column in 244 Fig. 4c) quickly dropped to almost zero, which is consistent with the continuity equation for 245 near-balanced non-divergent large-scale motions.

There are also similarities and differences in spectral slopes among different flight segments depicted in Fig. 4. For example, the above-mentioned spectral shapes of u and v from segment J3 are similar to those from segment J2 (i.e., the second and third columns in Fig. 4a and Fig 4b). Such consistent signals probably result from sampling under similar large-scale background flow at similar flight altitude with almost identical topography, especially between

251 the adjacent flight segments J1+J2 and J3. Despite the overall resemblance among the flight 252 segments of RF02, there are some unique characteristics in the power spectral distributions for 253 individual segments. For segments M1 and M2, for example, (i.e., the fourth column versus the 254 fifth column in Fig. 4), the slopes of u and v during segment M1 are approximately consistent 255 with a -3 power law for the scale of  $\sim 0.5 - 8$  km, while those during segment M2 follows a -5/3 256 power law instead. This is probably associated with the fact that segment M2 successfully 257 captures a rapid decrease in u (from ~65 m/s to ~40 m/s) while segment M1 has no such a 258 dramatic reduction in u (the fourth column in Fig. 3a versus the fifth column in Fig. 3a). Note 259 that the aircraft during segment M1 flew away from the jet core region, as the jet was still 260 moving eastward to the downhill side of the topography. In contrast, the aircraft during segment 261 M2 flew directly toward the approaching jet core at a lower flight level than segment M1 (the 262 fourth column in Fig. 3d versus the fifth column in Fig. 3d), and the observed decline of u (i.e., a 263 potential jet exit region) is located roughly on the downhill side of the topography (the fifth 264 column in Fig. 3d). This suggests that the spectral slopes for the aircraft measurements can, in 265 fact, be extremely sensitive to changes in the background flow, even though sampling takes place 266 in the same area only a few hours apart.

Figure 5 shows composite spectra for eight selected variables averaged over 68 flight segments. Unsurprisingly, the composite spectra are much smoother due to averaging. For *u* (Fig. 5a), *v* (Fig. 5b), and horizontal wind speed *V* (Fig. 5d), the slope of the power spectra are consistent with a -5/3 power law for scales above ~8-~16 km. For *w* (Fig. 5c), its spectral slope is generally consistent with -3 power laws for the scale of ~0.5-~2 km but is nearly zero for scales over 32 km, while the slopes in between (~2-~32 km) appear to follow an approximate -5/3 power law, with a statistically significant spectral peak at ~8-16 km. Even though the kinetic 274 energy spectra (Fig. 5e) may show a -5/3 slope that covers a larger range, the -3 slope over small 275 scale in KE is still evident. For  $\theta$  (Fig. 5f) at scales between ~0.5 km and ~2 km, its slope also 276 obeys a -3 power law. For  $\theta$  (Fig. 5f) at the scale greater than ~8-~16 km, the slope of power 277 spectrum tends to have a -5/3 slope, which is similar to u (Fig. 5a), v (Fig. 5b), and V (Fig. 5d) 278 for the same scales. For all the three pressure-related variables (i.e.,  $p_c$  in Fig. 5g,  $p_s$  in Fig. 5h,  $p_h$  in Fig. 5i), their slopes generally fall around a -5/3 power law, except for scales less than ~4 279 280 km in  $p_h$  (Fig. 5i). However, it is noteworthy that there is a sudden concavity (convexity) in  $p_c$  $(p_s \text{ or } p_h)$  for scales between ~4 km and ~16 km (also see the discussion in section 5.3). 281

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### 283 4. Wavelet analysis

# 284 *4.1 Single-variable wavelet analysis*

285 Standard spectral analysis methods characterize the variance as a function of wavelength 286 for an entire data record (flight segment), but do not indicate where variance of a particular 287 wavelength is located within the data record. We use wavelet analysis to complement the 288 spectral analysis in section 3 to study the variance as a function of wavelength within the five 289 flight segments from RF02. A Morlet wavelet function is employed in this study (e.g., Torrence 290 and Compo 1998; Zhang et al. 2001; Woods and Smith 2010). This is a continuous wavelet 291 transform that uses non-orthogonal complex wavelet functions comprising a plane wave 292 modulated by a Gaussian function (e.g., equation 1 in Torrence and Compo 1998):

293 
$$\psi_0(\eta) = \pi^{-1/4} e^{i\omega_0 \eta} e^{-\eta^2/2}$$
(2)

where  $\omega_0$  is the dimensionless wave number and  $\eta$  is the dimensionless distance. Here  $\omega_0$  is set to 6 to satisfy the admissibility condition (Farge 1992). The continuous wavelet transform, used to extract localized spectral information, is defined as the convolution of the series of interest xwith the complex conjugate of the wavelet (e.g., equation 2 in Torrence and Compo 1998)

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$$W_n(s) = \sum_{n'=0}^{N-1} x_{n'} \psi^* \left[ \frac{(n'-n)\Delta x}{s} \right]$$
(3)

where \* is the complex conjugate, *n* is the localized position index, *s* is the wavelet scale, and  $\Delta x$  is the resolution of the data (0.25 km in this case). The cone of influence (COI) shows the region of the wavelet spectrum where the edge errors cannot be ignored. Computation of the wavelet spectrum and edge error is performed with the wavelet function of equation 3 (Torrence and Compo 1998) in NCL.

304 Figure 6 contains the wavelet power spectra of five selected observed variables along the 305 five selected flight segments of RF02. Using the long segment J3 as an example again (third 306 column in Fig. 6), there is a substantial peak in the power of u (Fig. 6a) at wavelengths around 307 128-km between 400 and 700 km along the flight leg (also seen in  $p_c$  of Fig. 6e); ~100-km wave 308 power peaks at location 100-300 km; the wave power of wavelength from ~64 km to ~128 km 309 also peaks at location 1200-1400 km. The greatest similarity is between the spectra of w and  $\theta$ 310 (Figs. 6c and d). For example, from location 100 km to 800 km during segment J3, local 311 maximum of power in w (the third column in Fig. 6c) resembles the one in  $\theta$  (the third column 312 in Fig. 6d). In particular, three distinguished wave modes (~64 km, ~32 km, and ~10 km in 313 along-track wavelength) collocate at location 600-800 km (downstream of a localized hill around 314 600 km in the third column of Fig. 3d). Relatively persistent  $\sim 10$ -km waves in w are shown at 315 location 200-700 km, which corresponds to a similar peak in the spectral analysis of w in the 316 third column of Fig. 4c. Note that such ~10-km waves are also found in other flight segments in 317 RF02 (e.g., location 0-600 km during segment M1, the fourth column in Fig. 6c) and other 318 research flights in START08 (not shown). Interpretations of such small-scale localized wave
319 variances, as well as mesoscale localized wave variances, are discussed in section 5.

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# 321 *4.2 Polarization relationships from cross-wavelet analysis*

Following Woods and Smith (2010), the phase relationship between two variables (e.g., uand v, hereafter in short noted as  $(u'v')_p$ ) can be determined from the cospectrum  $(u'v')_c$  and quadrature spectrum  $(u'v')_q$ , which are defined as (also see section 6c in Torrence and Compo 1998; equation 8 and appendix A in Woods and Smith 2010):

326  $(u'v')_{c} = Re\{U_{n}(s_{i})V_{n}^{*}(s_{i})\}$ (4)

327 
$$(u'v')_q = Im\{U_n(s_j)V_n^*(s_j)\}$$
(5)

where  $U_n$  and  $V_n$  represent the wavelet transforms of u and v from equation 3,  $U_n(s_j)V_n^*(s_j)$  is 328 the complex-valued cross-wavelet spectrum, while  $Re\{\}$  and  $Im\{\}$  represent the real and 329 330 imaginary parts of the variables inside the parentheses, respectively. Woods and Smith (2010) focus on the energy flux by analyzing  $(p_c'w')_c$  from equation 4 for vertically propagating waves 331 and  $(p_c'w')_q$  from equation 5 for vertically trapped/ducted waves. In principle,  $(p_c'w')_p$  should 332 be, theoretically speaking, associated with  $(u'w')_p$   $((v'w')_p)$  (e.g., Eliassen and Palm 1960; 333 334 Lindzen 1990). This is particularly true for stationary mountain waves, which may be present for 335 RF02 given complex topography during each of the flight segments. However, in practice, 336 Woods and Smith (2010, their section 7) argued that the perturbation longitudinal velocity was 337 noisier than pressure in their study. In addition to equation 4 and equation 5, one can also define the absolute coherence phase angle as  $\frac{180}{\pi} \times \arctan\left(\left|\frac{Im\{U_n(s_j)V_n^*(s_j)\}}{Re\{U_n(s_j)V_n^*(s_j)\}}\right|\right)$  (also see section 6d in 338 339 Torrence and Compo 1998).

340 The phase relations among multiple variables are examined to further explore whether the 341 enhanced variances from the spectral and wavelet analyses are vertically propagating gravity waves. Figure 7 shows three selected examples of cospectrum analysis (i.e.,  $(u'w')_c$  in Fig. 7a, 342  $(v'w')_c$  in Fig. 7b,  $(p_c'w')_c$  in Fig. 7c), one selected example of quadrature spectrum analysis 343 (i.e.,  $(\theta'w')_q$  in Fig. 7d), and one example of absolute coherence phase angle for  $(\theta'w')_p$  (Fig. 344 345 7e). In the case of a single monochromatic internal gravity wave propagating vertically, for  $(u'w')_c$  (Fig. 7a), positive (negative) values indicate upward (downward) flux of along-track 346 momentum. For  $(v'w')_c$  (Fig. 7b), positive (negative) values indicate upward (downward) flux 347 of across-track momentum. For  $(p_c'w')_c$  (Fig. 7c), positive (negative) values indicate positive 348 (negative) vertical energy transport. For the quadrature spectrum of  $(\theta' w')_q$  (Fig. 7d), values 349 should be nonzero while the absolute coherence phase angle of  $(\theta'w')_p$  (Fig. 7e) should be close 350 351 to 90 degree.

352 We again take segment J3 as an example (the third column in Fig. 7): for the small-scale 353 component with along-track wavelength less than 50 km (horizontal solid line), enhanced but 354 incoherent variances are detected for location 100-500 km and for location 600-800 km, with fluctuating positive and negative values for both  $(u'w')_c$  (the third column in Fig. 7a) and 355  $(v'w')_c$  (the third column in Fig. 7b). The variations in the signs of vertical transports of 356 357 horizontal momentum fluxes imply that this flight segment is sampling waves propagating in 358 both forward and backward direction, assuming the vertical energy transports are generally upward. Correspondingly, the absolute coherence phase angle for  $(u'v')_p$  (not shown) also 359 360 alternates frequently between nearly 0 degree and nearly 90 degree. In particular, some of the 361 enhanced variances in the cospectra for along-track wavelengths from ~4 km to ~16 km, though 362 fluctuating in signs, are significant above the 95% confidence level.

For the mesoscale component with wavelengths from ~50 to ~100 km, remarkable localized quadrature variance is found in  $(\theta'w')_q$  (the third column in Fig. 7d) for location 500-800 km, consistent with the wavelet analysis of *w* in the third column of Fig. 6c and  $\theta$  in the third column of Fig 6d. The absolute coherence phase angle for  $(\theta'w')_p$  in Fig. 7e also demonstrate that the cross-wavelet spectrum between  $\theta$  and *w* is mostly dominated by their quadrature spectrum (red color shading in Fig. 7e), though there are some exceptions (blue color shading in Fig. 7e).

The similarities/discrepancies among different wavelet cospecta and quadrature spectra examined in Fig. 7 demonstrate the difficulties in gravity wave identification and the uncertainties in gravity wave characteristics estimation based solely on aircraft measurements.

In addition to cross-wavelet analysis, the signs of the net fluxes (e.g.,  $\overline{u'w'}$ ,  $\overline{v'w'}$ , and  $\overline{w'p_c}$ ) at each wavelength can also be estimated by the cospectrum analysis based on Fourier transform over the entire segment (not shown). Generally speaking, for the scale below ~32 km, both positive values and negative values are important in  $\overline{u'w'}$  and  $\overline{v'w'}$ , while positive  $\overline{w'p_c}$ appears to be more continuous than negative  $\overline{w'p_c}$ . For the scale above ~32 km, negative  $\overline{u'w'}$ (positive  $\overline{w'p_c}$ ) appears to be more continuous than positive  $\overline{u'w'}$  (negative  $\overline{w'p_c}$ ), while there is no dominant sign for  $\overline{v'w'}$  one way or the other.

380

### 381 5. Selected Wave-like Examples: signal of gravity waves or measurement noise?

382 This section examines several examples of wave-like variations during segment J3 in 383 more detail. Bandpass-filtered values of selected variables are computed by synthesizing the 384 wavelet transform using wavelets with scales between  $j_1$  and  $j_2$  using (e.g., equation 29 in 385 Torrence and Compo 1998)

386 
$$x_n' = \frac{\Delta j \Delta x^{1/2}}{C_\delta \psi_0(0)} \sum_{j=j_1}^{j_2} \frac{Re\{W_n(s_j)\}}{s_j^{1/2}} \quad (6)$$

where  $\Delta j$  is the scale resolution and  $C_{\delta}$  is a reconstruction factor taken as 0.776 for Morlet wavelet. The wavelet-based filter in equation 6 has the advantage in removing noise at each wave number and isolating single events with a broad power spectrum or multiple events with different wave number (Donoho and Johnstone 1994; Torrence and Compo 1998).

Nine pairs of variables, including  $(u'w')_p$ ,  $(v'w')_p$ ,  $(u'v')_p$ ,  $(p_c'u')_p$ ,  $(p_c'v')_p$ ,  $(p_c'w')_p$ ,  $(\theta'w')_p$ ,  $(p_s'w')_p$ , and  $(p_h'w')_p$ , are selected to examine whether the phase relationship of the variations in the airborne measurements is consistent with the linear theory for gravity waves. Generally speaking, the phase relation between two variables can be classified into two major categories: 1) In-phase or out-of-phase relationships, in which one variable leads or lags the other variable by approximately 0° or 180°; 2) Quadrature relationships, in which one variable leads or lags the other by approximately 90°.

The phase relationships for linear gravity waves are determined by theory and their 398 399 propagation characteristics. Take  $(u'w')_p$ ,  $(v'w')_p$ , and  $(p_c'w')_p$  as examples, if they have an 400 in- or out-of-phase relationship, the waves are propagating in the vertical direction; if they have a 401 quadrature relationship, the waves do not propagate vertically and may be trapped or ducted. Take  $(u'v')_p$  as another example, if they have an in- or out-of-phase relationship, the waves may 402 403 be internal gravity waves whose intrinsic frequencies are much higher than the Coriolis 404 frequency; if they have a quadrature relationship, the waves may be inertio-gravity waves with 405 intrinsic frequencies close to the Coriolis frequency. For vertically propagating linear gravity waves,  $(\theta'w')_p$  should have a quadrature relationship. According to Smith et al. (2008),  $p_h'$ 406

should dominate over  $p_s'$ , if the aircraft almost flies on a constant pressure surface. 407 Consequently,  $(p_h'w')_p$  should be almost identical to  $(p_c'w')_p$ . 408

- 409
- 410

# 5.1 Examples of mesoscale wave variances

411 Figure 8 demonstrates an example of potential mesoscale gravity waves selected based on the wavelet analysis of u (Fig. 6a), w (Fig. 6c),  $\theta$  (Fig. 6d), and  $p_c$  (Fig. 6e) for location 250-360 412 413 km in segment J3 (the exit region of northwesterly jet in Fig. 2d). The wave signals are further 414 highlighted by applying a wavelet-based filter (i.e., equation 6) to extract wavelike variations 415 with along-track wavelength between 100 and 120 km. Panels a, b, d, and e show out-of-phase relationships for  $(u'w')_p$ ,  $(v'w')_p$ ,  $(p_c'u')_p$ , and  $(p_c'v')_p$  respectively; while panels c, f, and i 416 show in-phase relationships for  $(u'v')_p$ ,  $(p_c'w')_p$ , and  $(p_h'w')_p$ . Panels g and h show 417 quadrature relationships for  $(\theta'w')_p$  and  $(p_s'w')_p$ . The observed phase relations shown in Fig. 8 418 419 are generally consistent with linear theory for propagating monochromatic gravity waves, as 420 indicated by the cospectrum/quadrature spectrum analysis in Fig. 7. These signals are likely to be internal gravity waves (due to the in-phase relation of  $(u'v')_p$  in Fig. 8c) with positive vertical 421 422 group velocity (due to their positive vertical energy flux, Fig. 8f).

In contrast, Figure 9 is an example of wave-like disturbances that lacks a clear, 423 424 propagating, linear-wave, phase relationship. This example is also selected based on the wavelet analysis of segment J3 for u, v, and  $p_c$  (Figs. 6a, b, and e) for along-track wavelength near 128 425 426 km and location between 560 and 688 km along the segment. This segment lies above the 427 complex topography as depicted in the third column of Fig. 3d. According to Figs. 9a-9e,  $(u'w')_p$ ,  $(u'v')_p$ , and  $(p_c'u')_p$  seem to have out-of-phase relationships, while  $(v'w')_p$  and 428  $(p_c'v')_p$  have almost perfect in-phase relationships. These phase relationships appear to be 429

430 reasonable and generally consistent with the linear theory. The near in-phase relationship exhibited by  $(\theta'w')_p$  (Fig. 9g), however, raises doubts about whether these variations are true 431 gravity waves, as this is not consistent with linear theory. If they are in fact gravity wave signals, 432 433 the discrepancy highlights the difficulties of extracting gravity wave perturbations from 434 observations. For example, the mesoscale variances may be contaminated by small-scale 435 variability of  $\theta$  and w due to the coexistence of wave variances at different scales for this region 436 (see the wavelet analysis of w in Fig. 6c in and  $\theta$  in Fig. 6d). Additionally, there are uncertainties 437 in extracting mesoscale gravity waves from a varying background flow (e.g., Zhang et al. 2004), 438 especially for u, v and  $\theta$ . Note that  $\theta$  and w have a very consistent quadrature relation from ~8 439 km to ~64 km for this region in their quadrature spectrum of Fig. 7d (also see Fig. 7e), but this 440 quadrature relation (the third column in Fig. 7d), including their corresponding wavelet spectrum 441 (the third column in Fig. 6c and Fig. 6d) is much weaker for wavelengths near 128 km for 442 location 560-688 km in segment J3.

443 Consistent with Smith et al. (2008), the amplitude of  $p_h'$  is much larger than the 444 amplitude of  $p_s'$  for both examples of mesoscale wave variances. Therefore,  $(p_h'w')_p$  is almost 445 identical to  $(p_c'w')_p$  for both cases (Fig. 8f versus Fig. 8i; Fig. 9f versus Fig. 9i). It appears that 446 the assumption of constant  $p_s$  flight height is valid for these two mesoscale examples.

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### 448 5.2 Examples of small-scale wavelike variations

Figure 10 shows an example of short-scale wave-like disturbances that have a phase relationship consistent with linear gravity wave theory based on the wavelet analysis in Fig. 6 with scales from 32 to 64 km located at 650 to 750 km during segment J3. In-phase relationships are seen in the filtered signals of  $(p_c'v')_p$  (Fig. 10e), while out-of-phase relationships are seen in 453  $(u'v')_p$  and  $(p_c'u')_p$  (Figs. 10c and d). Quadrature relationships can generally be seen in 454  $(u'w')_p$ ,  $(v'w')_p$ ,  $(p_c'w')_p$ , and  $(\theta'w')_p$  (Figs. 10a, b, f, and g). These small-scale waves have 455 no apparent vertical flux of horizontal momentum (Figs. 10a and b) and no vertical energy flux 456 (Fig. 10f), a key sign of vertically trapped gravity waves. Short-scale waves based on GV aircraft 457 measurements and/or numerical simulations are also discussed in Smith et al. (2008), Woods and 458 Smith (2010; 2011).

459 However, parts of the small-scale wave variations derived from the *in-situ* measurements, 460 especially for wavelengths from 5 to 15 km, may be difficult to classify as gravity waves. Figure 461 11 shows an example of short-scale wave variations in the aircraft measurements with along-462 track wavelengths from 8 to 16 km for locations 680 to 780 km along segment J3. As depicted in Fig. 11,  $(u'w')_p$  (Fig. 11a) appears to have a quadrature relationship, even though this relative 463 phase varies, especially for locations from 710 to 730 km. Compared to  $(u'w')_p$  (Fig. 11a), 464  $(v'w')_p$  and  $(\theta'w')_p$  (Fig. 11b and g) have consistent quadrature relationships within this 100-465 km distance. On the other hand,  $(u'v')_p$  (Fig. 11c) varies significantly from one wavelength to 466 467 the next. The amplitude of w' in this example is extremely large (~2.5 m/s at its maximum) in this selected example. In comparison, the amplitude of  $p_c'$  is rather small, and it is actually too 468 469 small to be noticed when using a wider bandpass window (not shown). Also, the quadrature relationship in  $(p_c'w')_p$  (Fig. 11f) is not as remarkable as those in  $(u'w')_p$  and  $(v'w')_p$  (Figs. 470 471 11a and b), which appears to contradict the theoretical description of Eliassen and Palm (1960) 472 on energy and momentum fluxes (also see Lindzen 1990). In addition, it is worth mentioning that  $(p_s'w')_p$  and  $(p_h'w')_p$  in Figs. 11h and i have almost perfect out-of-phase and in-phase 473 474 relationships, respectively.

In contradiction to Smith et al. (2008), the amplitude of  $p_h'$  in the above example of Fig. 11 is comparable with the amplitude of  $p_s'$  (Fig. 11h versus Fig. 11i). Surprisingly,  $(p_c'w')_p$ , ( $p_s'w')_p$ , and  $(p_h'w')_p$  are also very different from each other (compare Figs. 11f, h, and i). The signals of  $p_s'$  and  $p_h'$  (Fig. 11h and i) are out-of-phase for wavelengths near 10 km and have comparable amplitude, which leads to nearly no such wave variances in  $p_c'$  (Fig. 11d-11f) given  $p_c'$  is the sum of  $p_s'$  and  $p_h'$ .

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### 482 5.3 Insight from spectral analysis of different pressure variables

Figure 12a compares the power spectrum of three pressure-related variables (i.e., 483 484 corrected static pressure  $p_c$ , static pressure  $p_s$ , hydrostatic pressure correction  $p_h$ ; also see equation 1). Using segment J3 as an example, for wavelengths greater than  $\sim$ 32 km,  $p_c$  is almost 485 identical to  $p_h$ ; for wavelengths between ~32 km and ~4 km, the variances between  $p_s$  and  $p_h$ 486 487 are comparable, and the variances of  $p_c$  are noticeably smaller than those in  $p_s$  and  $p_h$ ; for wavelengths less than ~4 km,  $p_c$  is almost identical to  $p_s$ . Figure 12b shows the quantity 488  $\int \frac{spec(p_s)+spec(p_h)}{spec(p_c)}$ , where spec() indicates the power spectrum of the variable inside the 489 490 parentheses (e.g., Figs. 4-5). For segment J3, the square root of the ratio is close to 1.0 for the 491 wavelengths greater than ~32 km and less than ~4 km. At intermediate wavelengths, the square 492 root of the ratio reaches a maximum near 10 for wavelengths of ~10 km. This suggests that  $p_s'$ and  $p_h'$  may tend to cancel each other at intermediate scales, which reduces the amplitude of  $p_c'$ 493 at these intermediate wavelengths (also see the example in Fig. 11) since  $p_c'$  is the sum of  $p_s'$ 494 and  $p_h'$ . Similar behaviors can be also observed in other segments, although the exact ranges of 495 496 the intermediate wavelengths may be different from case to case.

Figure 12 suggests that the assumption of constant  $p_s$  flight height may not be valid at all scales, though it appears to be true for mesoscale waves. In consequence,  $p_h'$  may not always dominate over  $p_s'$  as assumed in Smith et al. (2008). The spectral analysis and wavelet analysis of  $p_s$  (not shown) demonstrate that  $p_s$  indeed has relatively large variances for the short-scale range, and that  $p_s$  and w share some common characteristics (also see Fig. 3). Moreover, the hydrostatic approximation, which is the underlying assumption for equation 1, may no longer be valid for short scales.

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# 505 6. Concluding remarks and discussion

506 One of the primary objectives of the recent START08 field experiment is to characterize 507 the sources and impacts of mesoscale waves with high-resolution flight-level aircraft 508 measurements and mesoscale models. The current study focuses on the second research flight 509 (RF02), which was the first airborne mission dedicated to probing gravity waves associated with 510 strong upper-tropospheric jet-front systems and high topography. Based on spectral and wavelet 511 analyses of the *in-situ* observations, along with a diagnosis of the polarization relationships, it is 512 found that there are clear signals of significant mesoscale variations with wavelengths ranging 513 from ~50 to ~500 km in almost every segment of the 8-hr flight (order ranging from 0.01 m/s to 514 1.0 m/s in vertical motion), which took place mostly in the lower stratosphere. The flow sampled 515 by the aircraft covers a wide range of background conditions including near the jet core, a jet 516 over the high mountains, and the exit region of the jet. In contrast, smaller-scale wavelike 517 oscillations below 50 km are found to be quite transient. In particular, aircraft measurements of 518 several flight segments are dominated by signals with sampled periods of ~20-~60 seconds and 519 wavelengths of ~5-~15 km (assuming that the typical flight speed is approximately 250 m/s).

520 This study suggests that at least part of the nearly-periodic high-frequency signals might 521 be unphysical and a result of intrinsic observational errors in the aircraft measurements or small-522 scale flight-altitude fluctuations that are difficult to account for. Such potentially contaminated 523 variations are often collocated with larger-scale wave signals, which in turn may lead to larger 524 uncertainties in the estimation of the wave characteristics. Part of the uncertainties may come 525 from the inability of the aircraft to maintain constant static pressure altitude in the presence of 526 small-scale turbulence. The current study mainly focuses on examining the fluctuations with the 527 use of linear theory for monochromatic gravity waves. Therefore, in addition to measurement 528 errors, the possibilities that those fluctuations may be due to other physical phenomena (e.g., 529 nonlinear dynamics, shear instability and/or turbulence) cannot be completely ruled out. 530 Nevertheless, despite the presence of possibly spurious wave oscillations in different flight 531 segments, the power spectra of horizontal winds and temperature averaged over many START08 532 flight segments generally follow the -5/3 power law. The common characteristics and individual 533 features of the wave variances and spectrum slope behaviors appear to be generally consistent 534 with past studies on the spectral analysis of aircraft measurement, including Nastrom and Gage 535 (1985) using the Global Atmospheric Sampling Program (GASP) flight dataset, and Lindborg 536 (1999) using the Measurement of Ozone and Water Vapor by Airbus In-Service Aircraft 537 (MOZAIC) aircraft observations. Spectral behaviors of atmospheric variables have also been 538 studied by high-resolution non-hydrostatic mesoscale numerical weather prediction (NWP) 539 models (e.g., Skamarock 2004; Tan et al. 2004; Zhang et al. 2007; Waite and Snyder 2013; Bei 540 and Zhang 2014).

541 Although the real-time mesoscale analysis and prediction system gave a reasonable 542 forecast guidance on the region of potential gravity wave activities, it remains to be explored (1)

how well the current generation of numerical weather models predicts the excitation of gravity waves, (2) how often gravity waves break in the ExUTLS region, and (3) what evidence in tracer measurements is shown for the contribution of gravity wave breaking to mixing. Future work will also seek to examine the origin and dynamics of the gravity waves observed during RF02 of START08 through a combination of observations and numerical modeling. This will help to distinguish whether the sampled mesoscale and small-scale variances are gravity waves or artifacts of the observing system.

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# 777 Figure Captions

778 Figure 1. The 68 Gulfstream V (GV) flight segments (colored lines) selected for wave analysis 779 during START08. The 18 colors represent 18 research flight (RF) missions. The thick blue lines 780 represent the second flight (RF02). The grey shadings give the terrain elevation map (shaded 781 every 250 m) over north America. The 4 black boxes are the model domain design for the second 782 research flight (RF02) during 21-22 April 2008, which are named D1-D4 from coarse to fine 783 domain with horizontal resolution as 45 km, 15 km, 5 km and 1.67 km, respectively. The field 784 of 18 RFs available online catalog the are (at 785 http://catalog.eol.ucar.edu/start\_08/missions/missions.html). The GV ground tracks of the 786 18 RFs are also documented in Fig. 2 of Pan et al. (2010).

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788 **Figure 2.** Simulated pressure at 9 km altitude (black contours; unit in hPa;  $\Delta = 2hPa$ ), horizontal wind speed at 9 km altitude (black shadings; unit in  $ms^{-1}$ ; levels at 30, 40, 50, 60 789 790  $ms^{-1}$ ), and the mesoscale component of horizontal divergence at 12.5 km (blue contours, 791 positive; red contour, negative; contour levels at  $\pm 7.5, \pm 15, \pm 30, \pm 60 \times 10^{-5} s^{-1}$ ) during RF02 792 in START08, with marked GV flight track (blue line) at selected time: (a) entire flight track at 21 793 April 18:00 UTC, (b) segment J1 at 21 April 19:10 UTC, (c) segment J2 at 21 April 19:50 UTC, 794 (d) segment J3 at 21 April 22:10 UTC, (e) segment M1 at 21 April 23:10 UTC, and (f) segment 795 M2 at 22 April 00:20 UTC. The triangle and circle marks represent the aircraft at the start time 796 of the segment and at selected time. The two-dimensional (2D) variables are based on D4 in Fig. 797 1. A band-pass filter is applied to extract signals with wavelength from 50 to 500 km for 798 horizontal divergence.

Figure 3. GV flight-level aircraft measurements during 5 selected segments (from left to right: 800 801 J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track velocity component (red; unit in  $ms^{-1}$ ; left y axis), across-track velocity component (blue; unit in  $ms^{-1}$ ; right y axis) and 802 803 horizontal velocity component (black; unit in  $ms^{-1}$ ; left y axis), (b) vertical velocity component (red; unit in  $ms^{-1}$ ; left y axis) and potential temperature (blue; unit in K; right y axis), (c) 804 805 perturbation of hydrostatic pressure correction (red; unit in hPa; left y axis), static pressure 806 (blue; unit in hPa; right y axis) and corrected static pressure (black; unit in hPa; left y axis), and 807 (d) flight height (red; unit in km; left y axis) and terrain (blue; black shading below terrain; unit 808 in km; right v axis). The series in segment J3 and M2 are reversed to facilitate the comparison 809 with J1+J2 and M1, respectively. Therefore, the orientation of x axis is from west to east along 810 each flight segment. The distance between minor tick marks in x axis is 100 km. The 811 perturbations in (c) are defined as the differences between the original data and their mean from 812 their corresponding segments.

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Figure 4. The spectrum (black line) of GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (b) across-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (c) vertical velocity component (unit:  $m^2s^{-2} \cdot m$ ), (d) potential temperature (unit:  $K^2 \cdot m$ ), and (e) corrected static pressure (unit:  $hPa^2 \cdot m$ ). Green lines show the theoretical Markov spectrum and the 5% and 95% confidence curves using the lag 1 autocorrelation. The blue (red) reference lines have slopes of -5/3 (-3).

822 Figure 5. Composite spectrum (black line) of GV flight-level aircraft measurement averaging 823 over all 68 segments in START08 (colored lines in Fig. 1): (a) along-track velocity component (unit:  $m^2 s^{-2} \bullet m$ ), (b) across-track velocity component (unit:  $m^2 s^{-2} \bullet m$ ), (c) vertical velocity 824 component (unit:  $m^2 s^{-2} \bullet m$ ), (d) horizontal velocity component (unit:  $m^2 s^{-2} \bullet m$ ), (f) potential 825 temperature (unit:  $K^2 \bullet m$ ), (g) corrected static pressure (unit:  $hPa^2 \bullet m$ ), (h) static pressure 826 (unit:  $hPa^2 \cdot m$ ), and (i) hydrostatic pressure correction (unit:  $hPa^2 \cdot m$ ). The subplot (e) kinetic 827 energy (unit:  $m^2 s^{-2} \cdot m$ ) is the sum of (a)-(c). Green lines show the composite curves of the 828 829 theoretical Markov spectrum and the 5% and 95% confidence curves using the lag 1 830 autocorrelation. The blue (red) reference lines have slopes of -5/3 (-3).

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832 Figure 6. Wavelet power spectrum of GV flight-level aircraft measurement during 5 selected 833 segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track 834 velocity component, (b) across-track velocity component, (c) vertical velocity component, (d) 835 potential temperature, and (e) corrected static pressure. Reference line (black line) shows the 836 cone of influence (COI), and the area outside COI is where edge error becomes important. Black 837 contour lines with dot shading represent 95% significance level based on a red noise background 838 (also see Torrence and Compo 1998; Woods and Smith 2010). The x axis is the same as in Fig. 839 3, including the reversal of segment J3 and M2.

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Figure 7. The wavelet cospectrum of (a)  $(u'w')_c$ , (b)  $(v'w')_c$ , (c)  $(p_c'w')_c$ , (d) the quadrature spectrum of  $(\theta'w')_q$ , and (e) the absolute coherence phase angle of  $(\theta'w')_p$  for GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08. Reference line (black line) shows the cone of influence (COI), and the area outside COI is where edge error becomes important. Black contour lines with dot shading
represent 95% significance level (also see Torrence and Compo 1998; Woods and Smith 2010).
The x axis is the same as in Fig. 3, including the reversal of segment J3 and M2. The horizontal
black line marks the scale of 50 km.

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850 Figure 8. A relatively good/clean example of mesoscale variations during segment J3 (location 851 250-360 km): (a) along-track velocity component (red; unit in m/s) and vertical velocity 852 component (blue; unit in m/s), (b) across-track velocity component (red; unit in m/s) and vertical 853 velocity component (blue; unit in m/s), (c) along-track velocity component (red; unit in m/s) and 854 across-track velocity component (blue; unit in m/s), (d) corrected static pressure (red; unit in 855 hPa) and along-track velocity component (blue; unit in m/s), (e) corrected static pressure (red; 856 unit in hPa) and across-track velocity component (blue; unit in m/s), (f) corrected static pressure 857 (red; unit in hPa) and vertical velocity component (blue; unit in m/s), (g) potential temperature 858 (red; unit in K) and vertical velocity component (blue; unit in m/s), (h) static pressure (red; unit 859 in hPa) and vertical velocity component (blue; unit in m/s), and (i) hydrostatic pressure 860 correction (red; unit in hPa) and vertical velocity component (blue; unit in m/s). A wavelet-based 861 band-pass filter is applied to extract signals with wavelength from 100 to 120 km for all the 862 above flight variables.

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Figure 9. Same as in Fig. 8, but for a relatively bad/noisy example of mesoscale variations
during segment J3 (location 560-688 km). The wavelet-based band-pass window is 118-138 km.

867	<b>Figure 10.</b> Same as in Fig. 8, but for a relatively good/clean example of smaller-scale variations
868	during segment J3 (location 650-750 km). The wavelet-based band-pass window is 32-64 km.
869	
870	Figure 11. Same as in Fig. 8, but for an example of smaller-scale variations during segment J3
871	(location 680-780 km). The wavelet-based band-pass window is 8-16 km.
872	
873	Figure 12. (a) The spectrum of corrected static pressure (black), static pressure (blue), and
874	hydrostatic pressure correction (red) based on GV flight-level aircraft measurement during 5
875	selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08. (b) The
876	spectrum of the square root ratio (see the text for its definition).
877	

Table 1: The aircraft statistic parameters of five selected flight segment in RF02 during the
START08 field campaign. Column 1-7 represent the name, the starting time (s), the ending time
(s), the averaged flight height (km), the averaged static pressure (hPa), the total distance (km),
and the averaged flight speed (m/s) of each selected flight segment.

Flight Segment	Start (s)	End (s)	Averaged Flight	Averaged Static	Distance	Averaged Flight
			Height (km)	Pressure (hPa)	(km)	Speed (m/s)
J1	2450	5000	11.8	196.9	685.74	268.92
J2	5170	8620	12.5	178.7	908.53	263.34
J3	9120	16850	13.1	162.1	1641.93	212.41
M1	17100	20630	12.6	178.5	950.46	269.25
M2	21500	26430	11.0	227.6	946.90	192.07



Figure 1. The 68 Gulfstream V (GV) flight segments (colored lines) selected for wave analysis during START08. The 18 colors represent 18 research flight (RF) missions. The thick blue lines represent the second flight (RF02). The grey shadings give the terrain elevation map (shaded every 250 m) over north America. The 4 black boxes are the model domain design for the second research flight (RF02) during 21-22 April 2008, which are named D1-D4 from coarse to fine domain with horizontal resolution as 45 km, 15 km, 5 km and 1.67 km, respectively. The field 18 catalog of the RFs are available online (at http://catalog.eol.ucar.edu/start 08/missions/missions.html). The GV ground tracks of the 18 RFs are also documented in Fig. 2 of Pan et al. (2010).



**Figure 2.** Simulated pressure at 9 km altitude (black contours; unit in hPa;  $\Delta = 2hPa$ ), horizontal wind speed at 9 km altitude (black shadings; unit in  $ms^{-1}$ ; levels at 30, 40, 50, 60  $ms^{-1}$ ), and the mesoscale component of horizontal divergence at 12.5 km (blue contours, positive; red contour, negative; contour levels at  $\pm 7.5$ ,  $\pm 15$ ,  $\pm 30$ ,  $\pm 60 \times 10^{-5}s^{-1}$ ) during RF02 in START08, with marked GV flight track (blue line) at selected time: (a) entire flight track at 21 April 18:00 UTC, (b) segment J1 at 21 April 19:10 UTC, (c) segment J2 at 21 April 19:50 UTC, (d) segment J3 at 21 April 22:10 UTC, (e) segment M1 at 21 April 23:10 UTC, and (f) segment M2 at 22 April 00:20 UTC. The triangle and circle marks represent the aircraft at the start time of the segment and at selected time. The two-dimensional (2D) variables are based on D4 in Fig. 1. A band-pass filter is applied to extract signals with wavelength from 50 to 500 km for horizontal divergence.



**Figure 3.** GV flight-level aircraft measurements during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track velocity component (red; unit in  $ms^{-1}$ ; left y axis), across-track velocity component (blue; unit in  $ms^{-1}$ ; right y axis) and horizontal velocity component (black; unit in  $ms^{-1}$ ; left y axis), (b) vertical velocity component (red; unit in  $ms^{-1}$ ; left y axis) and potential temperature (blue; unit in K; right y axis), (c) perturbation of hydrostatic pressure correction (red; unit in hPa; left y axis), static pressure (blue; unit in hPa; right y axis) and corrected static pressure (black; unit in hPa; left y axis), and (d) flight height (red; unit in km; left y axis) and terrain (blue; black shading below terrain; unit in km; right y axis). The series in segment J3 and M2 are reversed to facilitate the comparison with J1+J2 and M1, respectively. Therefore, the orientation of x axis is from west to east along each flight segment. The distance between minor tick marks in x axis is 100 km. The perturbations in (c) are defined as the differences between the original data and their mean from their corresponding segments.



**Figure 4.** The spectrum (black line) of GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (b) across-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (c) vertical velocity component (unit:  $m^2s^{-2} \cdot m$ ), (d) potential temperature (unit:  $K^2 \cdot m$ ), and (e) corrected static pressure (unit:  $hPa^2 \cdot m$ ). Green lines show the theoretical Markov spectrum and the 5% and 95% confidence curves using the lag 1 autocorrelation. The blue (red) reference lines have slopes of -5/3 (-3).



**Figure 5.** Composite spectrum (black line) of GV flight-level aircraft measurement averaging over all 68 segments in START08 (colored lines in Fig. 1): (a) along-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (b) across-track velocity component (unit:  $m^2s^{-2} \cdot m$ ), (c) vertical velocity component (unit:  $m^2s^{-2} \cdot m$ ), (d) horizontal velocity component (unit:  $m^2s^{-2} \cdot m$ ), (f) potential temperature (unit:  $K^2 \cdot m$ ), (g) corrected static pressure (unit:  $hPa^2 \cdot m$ ), (h) static pressure (unit:  $hPa^2 \cdot m$ ), and (i) hydrostatic pressure correction (unit:  $hPa^2 \cdot m$ ). The subplot (e) kinetic energy (unit:  $m^2s^{-2} \cdot m$ ) is the sum of (a)-(c). Green lines show the composite curves of the theoretical Markov spectrum and the 5% and 95% confidence curves using the lag 1 autocorrelation. The blue (red) reference lines have slopes of -5/3 (-3).



**Figure 6.** Wavelet power spectrum of GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08: (a) along-track velocity component, (b) across-track velocity component, (c) vertical velocity component, (d) potential temperature, and (e) corrected static pressure. Reference line (black line) shows the cone of influence (COI), and the area outside COI is where edge error becomes important. Black contour lines with dot shading represent 95% significance level based on a red noise background. The x axis is the same as in Fig. 3, including the reversal of segment J3 and M2.



**Figure 7.** The wavelet cospectrum of (a)  $(u'w')_c$ , (b)  $(v'w')_c$ , (c)  $(p_c'w')_c$ , (d) the quadrature spectrum of  $(\theta'w')_q$ , and (e) the absolute coherence phase angle of  $(\theta'w')_p$  for GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08. Reference line (black line) shows the cone of influence (COI), and the area outside COI is where edge error becomes important. Black contour lines with dot shading represent 95% significance level. The x axis is the same as in Fig. 3, including the reversal of segment J3 and M2. The horizontal black line marks the scale of 50 km.





**Figure 8.** A relatively good/clean example of mesoscale variations during segment J3 (location 250-360 km): (a) along-track velocity component (red; unit in m/s) and vertical velocity component (blue; unit in m/s), (b) across-track velocity component (red; unit in m/s) and vertical velocity component (blue; unit in m/s), (c) along-track velocity component (red; unit in m/s) and across-track velocity component (blue; unit in m/s), (d) corrected static pressure (red; unit in hPa) and along-track velocity component (blue; unit in m/s), (e) corrected static pressure (red; unit in hPa) and across-track velocity component (blue; unit in m/s), (f) corrected static pressure (red; unit in hPa) and across-track velocity component (blue; unit in m/s), (f) corrected static pressure (red; unit in hPa) and vertical velocity component (blue; unit in m/s), (g) potential temperature (red; unit in K) and vertical velocity component (blue; unit in m/s), (h) static pressure (red; unit in hPa) and vertical velocity component (blue; unit in m/s), and (i) hydrostatic pressure correction (red; unit in hPa) and vertical velocity component (blue; unit in m/s). A wavelet-based band-pass filter is applied to extract signals with wavelength from 100 to 120 km for all the above flight variables.



Segment J3 (distance: 560-688km; bandpass window: 118-138km)

**Figure 9.** Same as in Fig. 8, but for a relatively bad/noisy example of mesoscale variations during segment J3 (location 560-688 km). The wavelet-based band-pass window is 118-138 km.





**Figure 10.** Same as in Fig. 8, but for a relatively good/clean example of smaller-scale variations during segment J3 (location 650-750 km). The wavelet-based band-pass window is 32-64 km.



Segment J3 (distance: 680-780km; bandpass window: 8-16km)

**Figure 11.** Same as in Fig. 8, but for an example of smaller-scale variations during segment J3 (location 680-780 km). The wavelet-based band-pass window is 8-16 km.



**Figure 12.** (a) The spectrum of corrected static pressure (black), static pressure (blue), and hydrostatic pressure correction (red) based on GV flight-level aircraft measurement during 5 selected segments (from left to right: J1, J2, J3, M1 and M2) of RF02 in START08. (b) The spectrum of the square root ratio (see the text for its definition).